

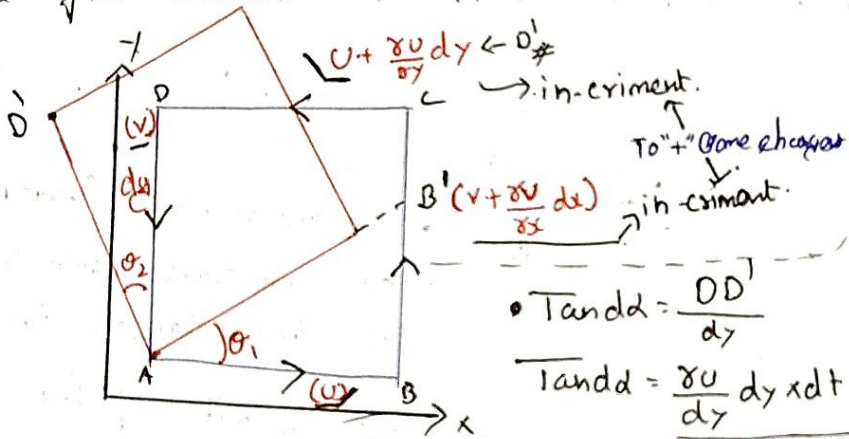
$\Rightarrow x = 0 \therefore$  non-rotation

$\Rightarrow x \neq 0 \therefore$  rotation

To Prove the equation

$$\omega = \omega_x i + \omega_y j + \omega_z k \quad \text{--- (1)}$$

The fluid element in  $x$  &  $y$  plane.



$$\Rightarrow D = v \times T$$

$$\tan d\alpha = \frac{BB'}{dx}$$

$$\tan d\alpha = \frac{\partial v}{\partial x} dx \times dt$$

$$\frac{d\alpha}{dt} = \frac{dv}{dx}$$

$$\omega_{AB} = \frac{\partial v}{\partial x}$$

$$\omega_{AD} = \frac{\partial u}{\partial y}$$

$$\frac{d\alpha}{dt} = \frac{du}{dy}$$

$$\omega_{AD} = \frac{\partial u}{\partial y}$$

$$\vec{\omega} = \frac{1}{2} \begin{vmatrix} \hat{i} & \hat{j} & \hat{k} \\ \frac{\partial}{\partial x} & \frac{\partial}{\partial y} & \frac{\partial}{\partial z} \\ u & v & w \end{vmatrix}$$

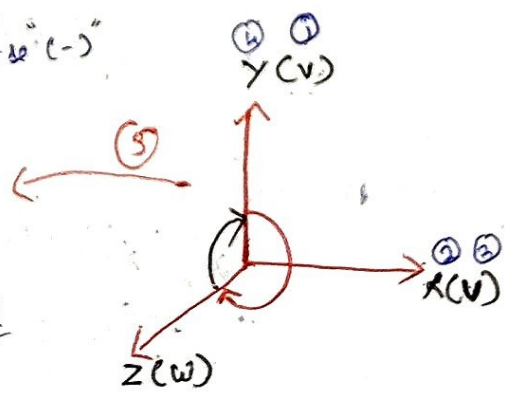
$$\omega_x = \frac{\omega_{AB} + \omega_{AD}}{2}$$

$$\omega_x = \frac{1}{2} \left[ \frac{\partial v}{\partial x} + \frac{\partial u}{\partial y} \right] \quad \text{--- (2)}$$

$$\omega_y = \frac{1}{2} \left[ \frac{\partial w}{\partial z} - \frac{\partial v}{\partial x} \right] \quad \text{--- (3)}$$

$$\omega_z = \frac{1}{2} \left[ \frac{\partial v}{\partial x} - \frac{\partial u}{\partial y} \right] \quad \text{--- (4)}$$

$$\omega_y = \frac{\omega_{AB} + \omega_{AD}}{2}$$



anti-clockwise (-)

but 2, 3, 4 on 1

eq 1:  $\omega = \omega_x i + \omega_y j + \omega_z k$

$\vec{\omega} = \frac{1}{2} \left[ \frac{\partial w}{\partial y} - \frac{\partial v}{\partial z} \right] \hat{j} + \frac{1}{2} \left[ \frac{\partial u}{\partial z} - \frac{\partial w}{\partial x} \right] \hat{j} + \frac{1}{2} \left[ \frac{\partial v}{\partial x} - \frac{\partial u}{\partial y} \right] \hat{k}$

Vorticity and Circulation

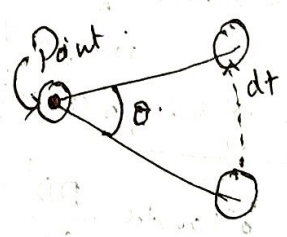
Vorticity ( $\Omega$ )

→ It indicates whether a <sup>→ air (or) water</sup> Particle is rotating or not.

⇒ angular velocity in general use and vorticity in fluid mechanics.

Angular velocity :-

⇒ It tells how much fast the angle changes when the object moves along a circular path.



$\omega = \frac{d\theta}{dt}$

- $\omega$  = angular velocity (in rad per sec)
- $d\theta$  = small change angular displacement
- $dt$  = small change in time

⇒ mathematically vorticity is twice the angle velocity of the Particle about its own axis.

$\Omega = 2\vec{\omega} \Rightarrow \textcircled{1}$

$\vec{\omega} = \frac{1}{2} \begin{vmatrix} \hat{i} & \hat{j} & \hat{k} \\ \frac{\partial u}{\partial x} & \frac{\partial v}{\partial y} & \frac{\partial w}{\partial z} \end{vmatrix} = \frac{1}{2} [\nabla \times \vec{V}]$

$\vec{\omega} = \frac{1}{2} [\nabla \times \vec{V}] \Rightarrow \textcircled{2}$

$\vec{\Omega} \Rightarrow$  EXIST  $\Rightarrow$  Rotational

$\vec{\Omega} \Rightarrow$  NOT EXIST  $\Rightarrow \hat{o}_1 + \hat{o}_2 + \hat{o}_3 \Rightarrow$  Ir-Rotational

Put 1 in eq 2

$\times \left[ \frac{1}{2} (\nabla \times \vec{V}) \right]$

$\vec{\Omega} = \nabla \times \vec{V}$

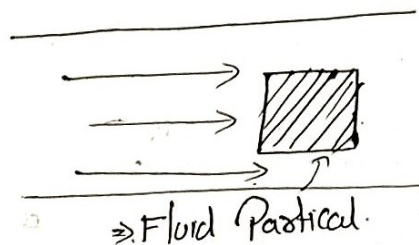
⇒ curl of velocity vector.

→ how much the field rotates around the points.

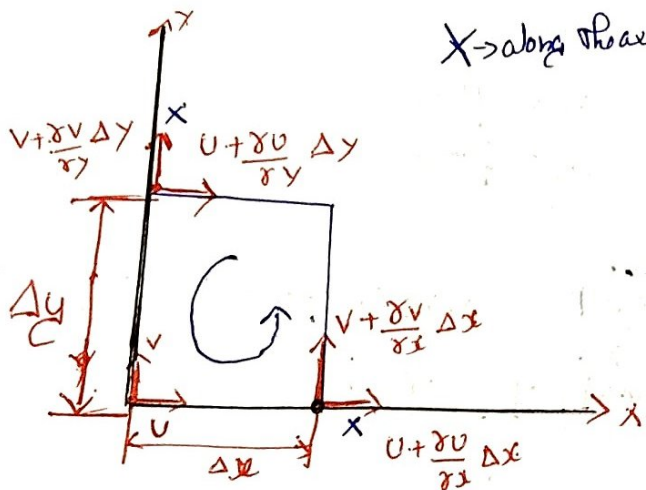
Circulation :-  $\Gamma$

$\Rightarrow$  line integral of tangential velocity component.

$\Rightarrow$  Circulation around a closed contour is sum of vorticity enclosed within it.



X  $\rightarrow$  along the axis.



$$\Gamma = \oint \vec{v} \cdot d\vec{s}$$

$$d\Gamma = (U \times \Delta x) - [v + \frac{\partial v}{\partial x} \Delta x] \Delta y - [u + \frac{\partial u}{\partial y} \Delta y] \Delta x - v \Delta y$$

$$= U \Delta x + v \Delta y + \frac{\partial v}{\partial x} \Delta x \Delta y - [v \Delta y + \frac{\partial v}{\partial x} \Delta x \Delta y] - [u \Delta x + \frac{\partial u}{\partial y} \Delta y \Delta x] - v \Delta y$$

$$= \left[ \frac{\partial v}{\partial x} - \frac{\partial u}{\partial y} \right] \Delta x \Delta y$$

$$= 2 \omega_z \Delta x \Delta y$$

$$2\omega_z = \Omega$$

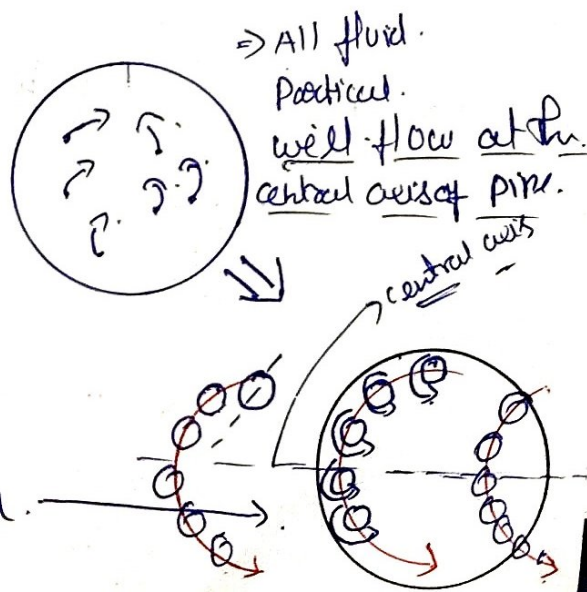
$$d\Gamma = 2\omega_z dA \rightarrow \text{differential circulation.}$$

$$\Gamma = \oint \vec{v} \cdot d\vec{s} = 2 \int \omega_z dA$$

$$\Gamma = \int \Omega dA$$

$\downarrow$  circulation       $\downarrow$  vorticity

$\Rightarrow$  circulation is sum of vorticity in the area.



# Stream function and Potential function.

→ etalon.  
Stream function :-  $(\psi) \rightarrow \text{PSI}$

⇒ A function to describe an incompressible flow

⇒ Introduced by Lagrange.

⇒ for 2D flow only.

⇒ Stream function  $\neq$  incompressible flow.  
 = incompressible flow.

⇒ P.D.  $\rightarrow \psi$   
 $u = \frac{\partial \psi}{\partial y}$  ;  $v = -\frac{\partial \psi}{\partial x}$   
 ⇒ giving in the velocity of x-direction.  
 ⇒ giving in the velocity of y-direction.

Continuity equation :-

$$\frac{\partial \rho}{\partial t} + \nabla(\rho \vec{v}) = 0 \Rightarrow \rho(\nabla \cdot \vec{v}) = 0 \Rightarrow \nabla \cdot \vec{v} = 0$$

$$\Rightarrow \frac{\partial u}{\partial x} + \frac{\partial v}{\partial y} = 0$$

$$\Rightarrow \frac{\partial}{\partial x} \left[ \frac{\partial \psi}{\partial y} \right] + \frac{\partial}{\partial y} \left[ -\frac{\partial \psi}{\partial x} \right] = 0$$

Uses of  $\psi$  :-

⇒ u and v are replaced by single variable  $(\psi)$ .

$$\psi \rightarrow \begin{cases} u = \frac{\partial \psi}{\partial y} \\ v = -\frac{\partial \psi}{\partial x} \end{cases}$$

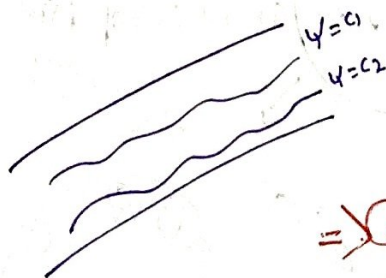
$$\Rightarrow \frac{\partial^2 \psi}{\partial x^2} - \frac{\partial^2 \psi}{\partial x^2} = 0$$

0 = 0  
 LHS = RHS.

∴ Continuity is satisfied  
 So flow exists.

⇒ Stream lines can be easily drawn.

eqn of stream line =  $\frac{dx}{u} = \frac{dy}{v}$



$$v dx - u dy = 0$$

$$-\frac{\partial \psi}{\partial x} dx - \frac{\partial \psi}{\partial y} dy = 0$$

$$\frac{\partial \psi}{\partial x} dx + \frac{\partial \psi}{\partial y} dy = 0$$

$$\oint \psi = 0$$

$$\int d\psi = \int 0 \Rightarrow \boxed{\psi = C}$$

⇒ Stream function is equal to a constant along the stream line.

Important point :-

•  $\psi \Rightarrow$  Continuity equation should be satisfied.

$\Downarrow$   
The flow is exist #

• Laplace eq<sup>n</sup> is satisfied  $\Leftrightarrow$  flow is irrotational of  $\psi$

of  $\psi$   
$$\nabla^2 \psi = \frac{\partial^2 \psi}{\partial x^2} + \frac{\partial^2 \psi}{\partial y^2} = 0 \Rightarrow$$
 Laplace eq<sup>n</sup> of  $\psi$

Potential function (or) velocity potential function ( $\phi$ )

$\phi \checkmark \rightarrow$  Ir-rot  
 $\phi \times \rightarrow$  rotat

$\Rightarrow$  A function to describe 3D irrotational flow.  
 $\Rightarrow$  Applicable for 2D as well as 3D flow.

Vector identity  $\Rightarrow$  Curl of divergence of a scalar fn = 0

Irrot  $\Rightarrow \left[ \begin{array}{l} \nabla \times (\nabla \phi) = 0 \\ \nabla \times \vec{V} = 0 \end{array} \right]$   
 $\vec{V} = \nabla \phi \rightarrow$  velocity potential function  
 $\rightarrow$  vorticity = 0  $\Rightarrow$  no rotation.

$u = \frac{\partial \phi}{\partial x} ; v = \frac{\partial \phi}{\partial y} ; w = \frac{\partial \phi}{\partial z}$

lines of  $\phi = c \Rightarrow$  equipotential lines

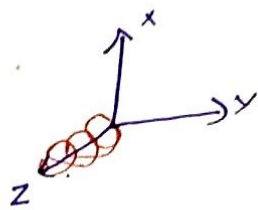
Important point :-

•  $\phi$  exist  $\Rightarrow$  flow is irrotational.

• Laplace eq<sup>n</sup> is satisfied  $\Rightarrow$  flow is possible.

### Problem

① A stream function is given by  $\psi = 3x^2 - y^3$ , Determine the magnitude of velocity components at the point (3, 1)



Sol:-

$$\Rightarrow \psi = 3x^2 - y^3$$

$$\Rightarrow (3, 1)$$

$x, y$

$\therefore$  They gave  $x, y$  so we want to find z-direction rotation.

$$\omega_z = \frac{1}{2} \left[ \frac{\partial v}{\partial x} - \frac{\partial u}{\partial y} \right] \rightarrow \text{①}$$

but eq ② & ③ on ①

$$\omega_z = \frac{1}{2} [-6 - (-6)] = 0$$

$$\boxed{\omega = 0}$$

$\therefore$  So, it is NO-rotational.

$$\frac{\partial \psi}{\partial y} = u$$

$$-\frac{\partial \psi}{\partial x} = v$$

$$\frac{\partial (3x^2 - y^3)}{\partial y} = u$$

$$\frac{\partial (3x^2 - y^3)}{\partial x} = v$$

$$-3y^2 = u$$

$$-6x = v$$

$$u = -3y^2 \rightarrow \text{②}$$

$$v = -6x \rightarrow \text{③}$$

$$\boxed{\vec{V} = -3y^2 \hat{i} - 6x \hat{j}}$$

velocity field.

For magnitude (3, 1)

$$\vec{V} = -3(1) \hat{i} - 6(3) \hat{j}$$

$$V = 3\hat{i} - 18\hat{j}$$

$$V = \sqrt{u^2 + v^2}$$

$$= \sqrt{3^2 + 18^2}$$

$$= \sqrt{9 + 324}$$

$$= \sqrt{333} \approx 3\sqrt{37}$$

$$= 18.24$$

Problem 1

In a two-dimensional, incompressible flow the fluid velocity components are given by  $u = x - 4y$ ;  $v = -y - 4x$  show that the flow satisfies the continuity equation and obtain the expression of the stream function. If the flow is potential (irrotational) obtain also the expression for the velocity potential.

Sol:-

$$u = x - 4y ; v = -y - 4x$$

$$\frac{\partial u}{\partial x} + \frac{\partial v}{\partial y}$$

$$-1 + 1 = 0 \quad \#$$

$$\frac{\partial \psi}{\partial y} = u \Rightarrow \int \partial \psi \Rightarrow \int u \, dy$$

$$\int d\psi = \int (x - 4y) \, dy$$

$$\int d\psi = xy - 2y^2 + f(x) + c_1$$

$\psi = xy - 2y^2 + f(x) + c_1 \quad \text{--- (1)}$

$$-\frac{\partial \psi}{\partial x} = v$$

$$\int d\psi = \int (y + 4x) \, dx$$

$\psi = xy + 2x^2 + f(y) + c_2 \quad \text{--- (2)}$

Compare eq (1) & (2)

$$\psi = xy + 2x^2 - 2y^2 + c \quad \#$$

$$\omega_z = \frac{1}{z} \left[ \frac{\partial v}{\partial x} - \frac{\partial u}{\partial y} \right] = 0$$

$$\omega_z = \frac{1}{2} [x - 4y - (-y - 4x)]$$

$\omega_z = 0 \quad \#$

$$\frac{\partial u}{\partial x}$$

$$\frac{\partial (x - 4y)}{\partial x}$$

$$\frac{\partial u}{\partial x} = 1$$

$$\frac{\partial v}{\partial y}$$

$$\frac{\partial (-y - 4x)}{\partial y}$$

$$\frac{\partial v}{\partial y} = -1$$


---


$$\frac{\partial \psi}{\partial y} = u$$

$$\frac{\partial (xy + 2x^2 - 2y^2)}{\partial y}$$

$x - 4y = u$

---


$$-\frac{\partial \psi}{\partial x} = v$$

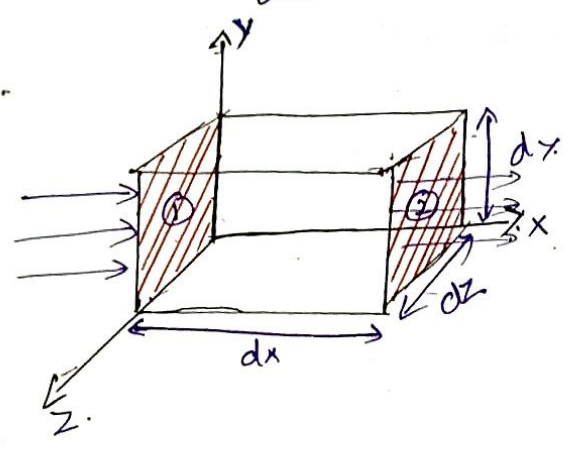
$$-\frac{\partial (xy + 2x^2 - 2y^2)}{\partial x}$$

$-y - 4x = v$

Module-3

Differential and kinetic of Continuity Equation

Formula:- 
$$\overset{\text{In}}{\downarrow} \dot{M}_1 - \overset{\text{out}}{\downarrow} \dot{M}_2 = \overset{\text{Net}}{\downarrow} \frac{dm}{dt}$$



x-direction:- 
$$\overset{\text{In}}{\downarrow} \rho_0 dy dz v - \overset{\text{out}}{\downarrow} \left[ \rho_0 + \frac{\partial \rho_0}{\partial x} dx \right] dy dz = \overset{\text{Net}}{\downarrow} - \frac{\partial \rho_0}{\partial x} dx dy dz \neq$$

y-direction:- 
$$- \frac{\partial \rho_0}{\partial y} dx dy dz \neq$$

z-direction:- 
$$- \frac{\partial \rho_0}{\partial z} dx dy dz \neq$$

$$\dot{M}_1 - \dot{M}_2 = \frac{d\rho}{dt} dx dy dz$$

$$\left[ -\frac{\partial \rho_0}{\partial x} - \frac{\partial \rho_0}{\partial y} - \frac{\partial \rho_0}{\partial z} \right] dx dy dz = \frac{d}{dt} (\rho_0 dx dy dz)$$

$$\frac{\partial \rho}{\partial t} + \frac{\partial \rho v}{\partial x} + \frac{\partial \rho v}{\partial y} + \frac{\partial \rho w}{\partial z}$$

Total net flow in unit time = change of mass flow in control volume in unit.

$$\frac{\partial \rho_0}{\partial x} + \frac{\partial \rho_0}{\partial y} + \frac{\partial \rho_0}{\partial z} = -\frac{\partial \rho}{\partial t} \Rightarrow \text{un-steady case} \neq$$

In-compressible:-

$\Rightarrow$  density will const. cancel

$$\frac{\partial v}{\partial x} + \frac{\partial v}{\partial y} + \frac{\partial w}{\partial z} = 0$$

Steady flow:-

$\Rightarrow$  time will cancel.

$$\frac{\partial \rho v}{\partial x} + \frac{\partial \rho v}{\partial y} + \frac{\partial \rho w}{\partial z} = 0$$

# Bernoulli's

⇒ it comes from applying The law of conservation of energy to a moving fluid.

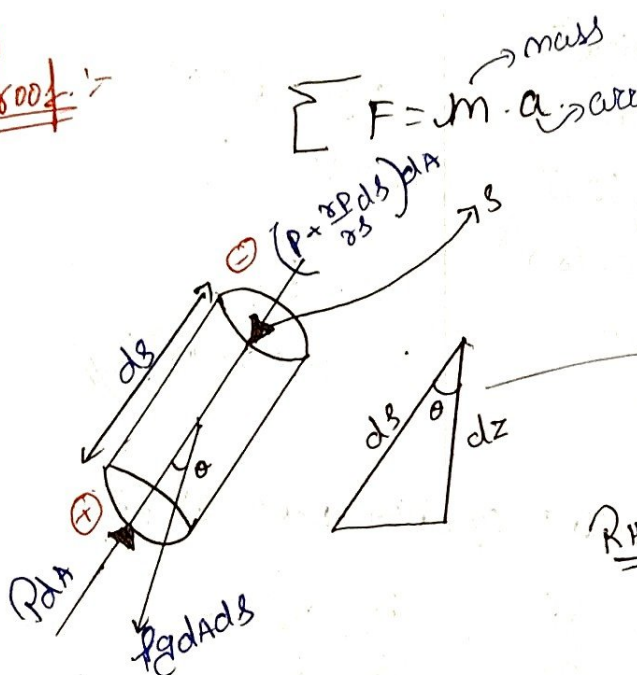
⇒ The general form of Bernoulli's equation (along a streamline, for an incompressible, non-viscous, steady flow, neglecting heat/work interaction.

$$\frac{P}{\rho g} + \frac{v^2}{2g} + z = \text{constant}$$

$\downarrow$  P.E       $\downarrow$  K.E       $\downarrow$  Spave.

- Internal friction. ✗
- Pressure force. ✓
- Shear force. ✗
- Gravitation force. ✓

## Proof:



$$\sum F = m \cdot a$$

$\rightarrow$  mass       $\rightarrow$  acceleration.

Pressure force is zero.

$$\sum F = +P \cdot dA - \left[ P + \frac{\partial P}{\partial s} ds \right] dA - \rho g dA ds \cos \theta$$

$$\cos \theta = \frac{dz}{ds}$$

$$\sum F = -\frac{\partial P}{\partial s} (dA ds) - \rho g (dA ds) \frac{dz}{ds} = m \cdot a$$

RHS :=

$$\rightarrow m \cdot a$$

$$\rightarrow \rho (dA \cdot ds) \frac{dv}{dt} \times \frac{ds}{ds}$$

$$\rightarrow \rho (dA \cdot ds) v \cdot \frac{dv}{ds} \neq$$

to remove time because it's steady flow.

$$= -\frac{\partial P}{\partial s} (dA ds) - \rho g (dA ds) \frac{dz}{ds} = \rho (dA ds) v \cdot \frac{dv}{ds}$$

$$\frac{\partial P}{\partial s} = \frac{dP}{ds}$$

change partial to finite differential.

IT is called eq<sup>n</sup> of Euler's eq<sup>n</sup> of the motion:-

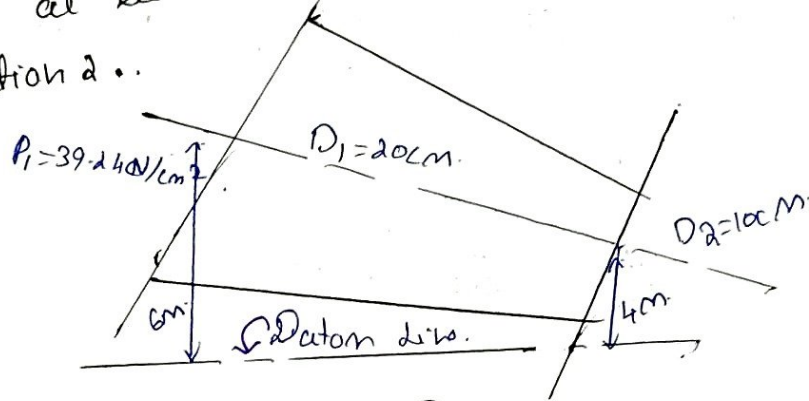
$$\Rightarrow -\frac{dP}{ds} (dA ds) - \rho g (dA ds) \frac{dz}{ds} = \rho (dA ds) v \frac{dv}{ds}$$

by using integration by parts.

$$\boxed{\frac{dP}{\rho} + g dz + v dv = 0}$$

$$\frac{P}{\rho g} + \frac{v^2}{2g} + z = C$$

① The water is flowing through pipe having diameter 20cm at section 1 and 10cm at section 2 respectively. The rate of flow through pipe is 35 litres/s. The section 1 is 6m above datum and section 2 is 4m above datum. If the pressure at section 1 is 39.24 N/cm<sup>2</sup>, find the pressure at section 2..



Given:-

Section ①

$D_1 = 20\text{cm} \Rightarrow 0.2\text{m}$

$A_1 = \frac{\pi}{4} (0.2)^2 \Rightarrow 0.0314\text{m}^2$

$P_1 = 39.24\text{ N/cm}^2$

$= 39.24 \times 10^4\text{ N/m}^2$

$z_1 = 6\text{m}$

Section ②

$D_2 = 0.10\text{m}$

$A_2 = \frac{\pi}{4} (0.1)^2$

$= 0.0078\text{m}^2$

$z_2 = 4\text{m}$

$P_2 = ?$

Rate flow

$Q = 35\text{ lit/s}$

$= \frac{35}{1000}$

$= 0.035\text{ m}^3/\text{s}$

Now,  $Q = A_1 v_1 = A_2 v_2$

Continuity equation.

$v_1 = \frac{Q}{A_1} = \frac{0.035}{0.0314} = 1.114\text{ m/s}$

$v_2 = \frac{Q}{A_2} = \frac{0.035}{0.00785} = 4.456\text{ m/s}$

Applying Bernoulli's equation section ① & ②

$$\frac{P_1}{\rho g} + \frac{v_1^2}{2g} + z_1 = \frac{P_2}{\rho g} + \frac{v_2^2}{2g} + z_2$$

$$\frac{39.24 \times 10^4}{1000 \times 9.81} + \frac{(1.114)^2}{2 \times 9.81} + 6.0 = \frac{P_2}{1000 \times 9.81} + \frac{(4.456)^2}{2 \times 9.81} + 4.0$$

$$40 + 0.063 + 6.0 = \frac{P_2}{9810} + 1.012 + 4.0$$

$$46.063 = \frac{P_2}{9810} + 5.012$$

$$\frac{P_2}{9810} = 46.063 - 5.012 \Rightarrow 41.051$$

$$P_2 = 41.051 \times 9810\text{ N/m}^2$$

$$P_2 = \frac{41.051 \times 9810}{10^4}\text{ N/cm}^2$$

$$= 40.27\text{ N/cm}^2$$

2) The water is flowing through a taper pipe of length 100m having (1) diameters 600mm at the upper end and 300mm at a lower end, at the rate of 50 liters/s. The pipe has a slope of 1 in 30. Find the pressure at the lower end if the pressure at the higher level is  $19.62 \text{ N/cm}^2$ .

Sol:-

length of pipe,  
 $\Rightarrow L = 100 \text{ m}$ .

• Dia. upper end,  $D_1 = 600 \text{ mm} = 0.6 \text{ m}$

$$\rightarrow \text{Area, } A_1 = \frac{\pi}{4} D_1^2 \Rightarrow \frac{\pi}{4} (0.6)^2$$

$$\Rightarrow 0.2827 \text{ m}^2$$

$P_1$  at upper end

$$\Rightarrow 19.62 \text{ N/cm}^2$$

$$\Rightarrow 19.62 \times 10^4 \text{ N/m}^2$$

• Dia. lower end,  $D_2 = 300 \text{ mm} = 0.3 \text{ m}$ .

$$\rightarrow \text{Area, } A_2 = \frac{\pi}{4} D_2^2 \Rightarrow \frac{\pi}{4} (0.3)^2$$

$$\Rightarrow 0.07068 \text{ m}^2$$

•  $Q = \text{rate of flow} \Rightarrow 50 \text{ liters/s} \Rightarrow \frac{50}{1000} = 0.05 \text{ m}^3/\text{s}$

Let the datum line passes through the centre of the lower end.

$$\boxed{Z_2 = 0}$$

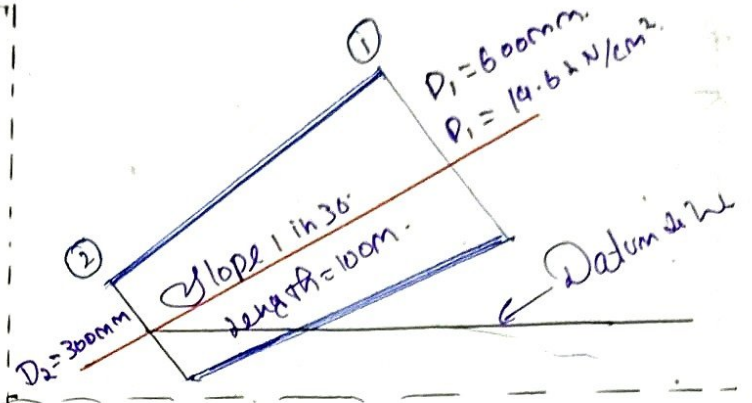
Also slope is 1 to 30 means

$$Z_1 = \frac{1}{30} \times 100 = \frac{10}{3} \text{ m}$$

Also we know  $\boxed{Q = A_1 V_1 = A_2 V_2}$

$$V_1 = \frac{Q}{A} \Rightarrow \frac{0.05}{0.2827} \Rightarrow 0.1768 \text{ m/sec} \Rightarrow 0.177 \text{ m/s}$$

$$V_2 = \frac{Q}{A} = \frac{0.05}{0.07068} = 0.7074 \text{ m/sec} \Rightarrow 0.707 \text{ m/s}$$



by applying Bernoulli's equation.

$$\frac{P_1}{\rho g} + \frac{V_1^2}{2g} + Z_1 = \frac{P_2}{\rho g} + \frac{V_2^2}{2g} + Z_2$$

$$\frac{19.62 \times 10^4}{1000 \times 9.81} + \frac{0.177^2}{2 \times 9.81} + \frac{10}{3} = \frac{P_2}{\rho g} + \frac{0.707^2}{2 \times 9.81} + 0$$

$$20 + 0.001596 + 3.334 = \frac{P_2}{\rho g} + 0.0254$$

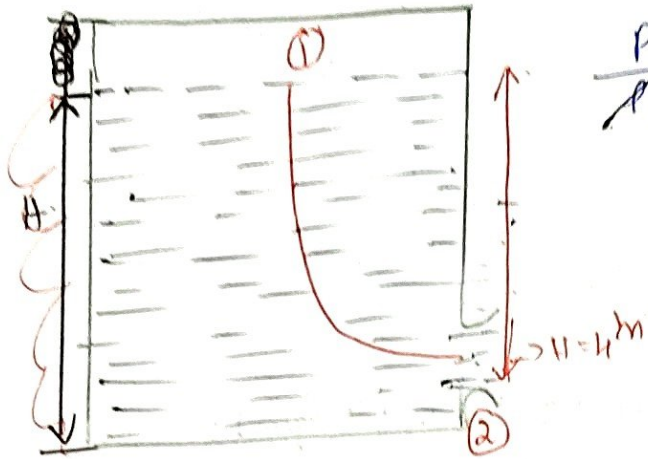
$$23.335 - 0.0254 = \frac{P_2}{1000 \times 9.81}$$

$$P_2 = 23.3 \times 9810 \text{ N/m}^2$$

$$= \frac{228573 \text{ N/m}^2}{10^4}$$

$$\Rightarrow 22.857 \text{ N/cm}^2 \#$$

Determine the velocity and mass flow rate of efflux from the circular hole (0.1 m dia) at the bottom of the water tank (at this instant). The tank is open to the atmosphere, and  $H = 4$  m.



$$\frac{P_1}{\rho g} + \frac{V_1^2}{2g} + z_1 = \frac{P_2}{\rho g} + \frac{V_2^2}{2g} + z_2$$

$$P_1 = P_2 = P_{atm}$$

$$V_1 \ll V_2 \quad (\text{or}) \quad V_1 \approx 0$$

$$z_1 = H$$

$$z_2 = 0$$

$$H = \frac{V_2^2}{2g}$$

$$V_2 = \sqrt{2gH}$$

velocity  
 Section 2:

$$= \sqrt{2 \times 9.81 \times 4}$$

$$= \sqrt{78.48}$$

$$= 8.85 \text{ m/s}$$

mass flow rate:

$$\dot{m} = \rho A V$$

→ diameter

$$= 1000 \times \frac{\pi}{4} (0.1)^2 (8.85)$$

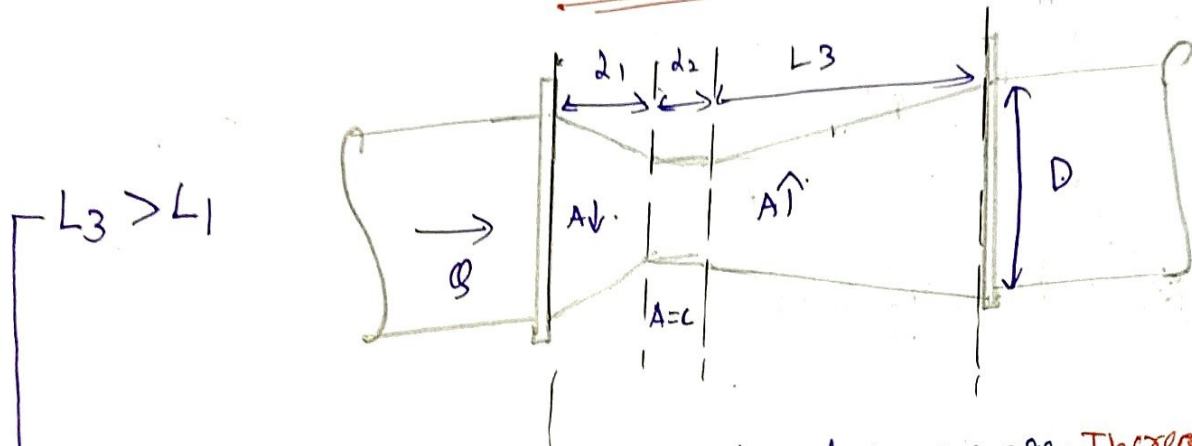
$$= 69.5 \text{ (kg/s)}$$

Application

- 1. Venturimeter.
  - 2. orifice meter.
  - 3. Pitot tube.
- } flow rate in pipe.

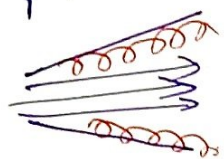
7.12M

Venturimeter



$L_3 > L_1$

• To avoid loss of pressure in boundary we are increasing the length of  $(d_3)$ .



region (1)

$\Rightarrow A \downarrow$  velocity  $(v) \uparrow$   
 $\rho A v = C$  :- Continuity equation.  
 $A_1 \downarrow v_1 \uparrow = C = A_2 v_2$

bernoules equation.

$$\frac{P}{\rho g} + \frac{v^2}{2g} = C$$

$\downarrow$                        $\downarrow$  = 1v

$A \downarrow v \uparrow P \downarrow$

region (2)

$\Rightarrow A \uparrow$  velocity  $(v) \downarrow$   
 $\rho A v = C$   
 $A_3 \uparrow v_3 \downarrow = C$

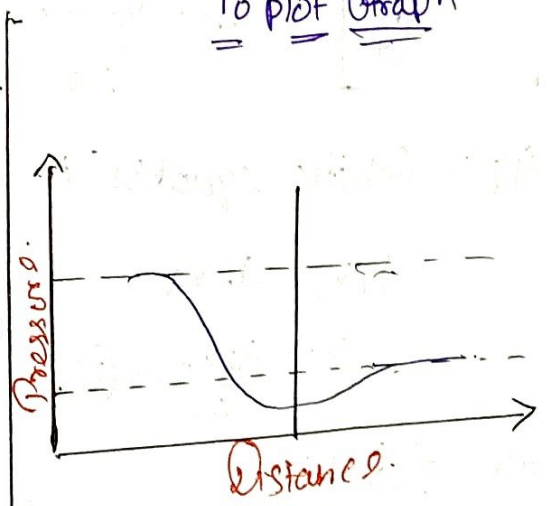
by bernoules equation.

$$\frac{P}{\rho g} + \frac{v^2}{2g} = C$$

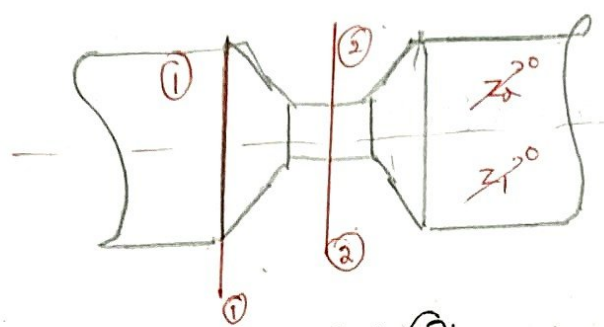
$\downarrow$                        $\downarrow$  = 1v

$A \uparrow v \downarrow P \uparrow$

To plot Graph



To measure flow rate through venturimeter.



by apply Bernoules eq. b/w ① & ②

$$\frac{P}{\rho g} + \frac{v^2}{2g} = C$$

$$\frac{P_1}{\rho g} + \frac{v_1^2}{2g} + z_1 = \frac{P_2}{\rho g} + \frac{v_2^2}{2g} + z_2$$

known value

$$\frac{P_1 - P_2}{\rho g} = \frac{v_2^2 - v_1^2}{2g} \rightarrow \text{①}$$

$$h = \frac{P_1 - P_2}{\rho g}$$

$$h = \frac{v_2^2 - v_1^2}{2g} \rightarrow \text{②}$$

Apply Continuity equation in section ① & ②

$$A_1 v_1 = A_2 v_2$$

$$v_1 = \frac{A_2 v_2}{A_1} \rightarrow \text{③}$$

but eq ③ in ②

$$h = \frac{v_2^2 - \left(\frac{A_2 v_2}{A_1}\right)^2}{2g}$$

$$2gh = v_2^2 \left[ 1 - \left(\frac{A_2}{A_1}\right)^2 \right]$$

$$2gh = v_2^2 \cdot \left[ \frac{A_1^2 - A_2^2}{A_1^2} \right]$$

$$v_2 = \frac{A_1 \sqrt{2gh}}{\sqrt{A_1^2 - A_2^2}}$$

$$\Rightarrow Q = A_1 v_1 = A_2 v_2 \Rightarrow$$

$$Q_{\text{theor}} = A_2 v_2 = \frac{A_1 A_2 \sqrt{2gh}}{\sqrt{A_1^2 - A_2^2}}$$

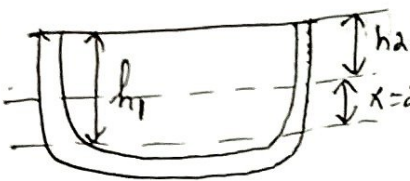
$$\frac{Q_{\text{actual}}}{Q_{\text{theor}}} = C_d$$

Discharge

Q. Oil of specific gravity 0.8 is flowing through a venturimeter having inlet diameter 20cm and throat diameter 10cm. The oil mercury difference manometer shows a reading of 25cm. Calculate the discharge of oil through the horizontal venturimeter. Take  $C_d = 0.98$

Given:-

$$Q_{th} = \frac{A_1 A_2 \sqrt{2gh}}{\sqrt{A_1^2 - A_2^2}}$$



$$x = 2.5 \text{ cm} \Rightarrow \frac{2.5}{100} = 0.025$$

Then given here mercury (heavy)

$$h = \frac{P_1 - P_2}{\rho g} = \frac{h_1 - h_2}{x} \left[ \frac{\rho_m}{\rho_o} - 1 \right]$$

oil (light)

$$h = 0.25 \left[ \frac{13.6}{0.8} - 1 \right]$$

$$h = 4 \text{ m of oil}$$

- 20cm  $\Rightarrow$  0.20m
- 10cm  $\Rightarrow$  0.10m
- 2.5cm  $\Rightarrow$  0.025

$$d_1 = 20 \text{ cm}$$

$$a_1 = \frac{\pi}{4} \times d_1^2 \Rightarrow \frac{\pi}{4} \times 20^2 = 314.16 \text{ cm}^2 \Rightarrow \frac{\pi}{4} \times (0.20)^2 = 0.0314 \text{ m}^2$$

$$a_2 = \frac{\pi}{4} \times d_2^2 \Rightarrow \frac{\pi}{4} \times 10^2 = 78.54 \text{ cm}^2 \Rightarrow \frac{\pi}{4} \times (0.10)^2 = 0.00785 \text{ m}^2$$

$$Q = C_d \cdot \frac{A_1 A_2}{\sqrt{A_1^2 - A_2^2}} \times \sqrt{2gh}$$

$0.98 \times \frac{0.0314 \times 0.00785}{\sqrt{0.0314^2 - 0.00785^2}} \times \sqrt{2 \times 9.81 \times 4.0} \Rightarrow \sqrt{78.48} \Rightarrow 8.86$

$$= 0.98 \times (0.0314^2 - 0.00785^2) = (0.000986 - 0.0000617) = 0.000924 \text{ m}^4$$

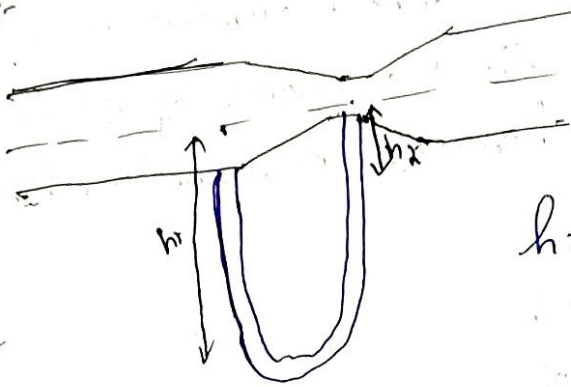
$$\frac{A_1 A_2}{\sqrt{A_1^2 - A_2^2}} \Rightarrow \frac{0.000247}{\sqrt{0.000924}} \Rightarrow \frac{0.000247}{0.0304} = 0.00813$$

$$Q = 0.98 \times 0.00813 \times 8.86$$

$$Q = 0.98 \times 0.0720 \Rightarrow Q = 0.0706 \text{ m}^3/\text{s}$$

$\Rightarrow$  71 liters/sec

Case - I :-



$$x = h_1 - h_2$$

specific gravity (Cheavy)

$$h = x \left[ \frac{\rho_h}{\rho_o} - 1 \right] \rightarrow \text{fluid passing to pipe}$$

Case - II :- (Invert manometer)

$$h = x \left[ 1 - \frac{\rho_1}{\rho_o} \right] \rightarrow \text{fluid passing to pipe}$$

lighter

Case - III :- vertical (higher pressure)

$$\frac{P_1}{\rho g} + \frac{V_1^2}{2g} + z_1 = P_2 + \frac{V_2^2}{2g} + z_2$$

$$\left[ \frac{P_1 - P_2}{\rho g} + z_1 - z_2 \right] = \frac{V_2^2 - V_1^2}{2g}$$

$$h = \left[ \frac{P_1}{\rho g} + z_1 \right] - \left[ \frac{P_2}{\rho g} + z_2 \right] = x \left[ \frac{\rho_h}{\rho_o} - 1 \right]$$

heavier

Case - IV

$$h = \left[ \frac{P_1}{\rho g} + z_1 \right] - \left[ \frac{P_2}{\rho g} + z_2 \right] = x \left[ \frac{\rho_h}{\rho_o} - 1 \right]$$

$$\left[ 1 - \frac{\rho_1}{\rho_o} \right] \rightarrow \text{lighter}$$

Problem

Find the discharge of water flowing through a pipe 30cm diameter placed as an venturi meter is inserted, 15 cm. The difference of Pressure between the main and through is measured by a liquid of sp. gr. 0.6 in an inverted U-tube between the main and throat is 30cm. The loss of head between the main and throat is 0.2 times the kinetic head of Pipe.

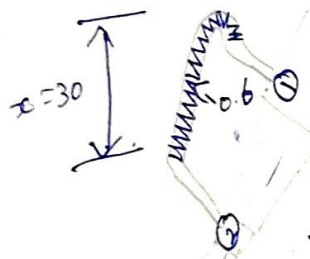
Sol:-

In question they not mention Pressure value, so we want to consider height @ formula of that

$$\frac{P_1 - P_2}{\rho g} + z_1 - z_2 = h = x \left[ 1 - \frac{\rho_l}{\rho} \right]$$

$$h = 0.3 \left[ 1 - \frac{0.6}{1} \right]$$

$$h = 0.12m$$



The loss are there so, apply the Bernoules in section 1 & 2 @ modify according to loss in head.

$$\left[ \frac{P_1}{\rho g} + \frac{v_1^2}{2g} + z_1 \right]_{\text{Total energy section 1}} = \left[ \frac{P_2}{\rho g} + \frac{v_2^2}{2g} + z_2 \right]_{\text{total energy section 2}} \rightarrow \text{No loss formula.}$$

$$(TE)_1 - (TE)_2 = 0.2 \times \frac{v_1^2}{2g}$$

$$\left[ \frac{P_1}{\rho g} + \frac{v_1^2}{2g} + z_1 \right] = \left[ \frac{P_2}{\rho g} + \frac{v_2^2}{2g} + z_2 + \frac{0.2 v_1^2}{2g} \right] \rightarrow \text{with loss}$$

$$\frac{P_1 - P_2}{\rho g} + z_1 - z_2 = \frac{V_2^2 - V_1^2}{2g} + \frac{0.2 V_1^2}{2g}$$

$$0.12 \text{ m} = \frac{V_2^2 - V_1^2}{2g} + 0.2 \frac{V_1^2}{2g} \rightarrow (1)$$

by using continuity equation:

$$A_1 V_1 = A_2 V_2$$

$$\frac{\pi}{4} (0.3)^2 \times V_1 = \frac{\pi}{4} (0.15)^2 \times V_2$$

$$V_2 = 4 V_1 \rightarrow (2) \Rightarrow V_1 = \frac{V_2}{4} \Rightarrow (2)$$

by sub (2) in (1)

$$0.12 + \frac{V_2^2}{2g} \left[ \frac{0.8}{16} - 1 \right] = 0$$

$$\frac{V_2^2}{2g} [0.05 - 1] = -0.12 \times 0$$

$$V_2 = \sqrt{\frac{2.481 \times 12}{0.95}}$$

$$V_2 = 157.4 \text{ cm} \Rightarrow 0.2 V_2$$

$$\Rightarrow 176.7 \times 157.4$$

$$\Rightarrow 27800 \text{ cm}^3/\text{s}$$

$$\Rightarrow 27.8 \text{ lit/s} \neq$$

(or)

$$0.12 = \frac{V_2^2 - \frac{1}{16} V_2^2}{2g} + \frac{0.2 V_2^2}{2g} \Rightarrow \frac{(1 - 0.0625 + 0.2) V_2^2}{2g}$$

$$1 - 0.0625 + 0.2 = 1.1375$$

$$0.12 = \frac{1.1375 V_2^2}{2g} \Rightarrow V_2^2 = \frac{0.12 \times 2 \times 9.81}{1.1375}$$

$$V_2^2 = \frac{2.3544}{1.1375} = 2.0716$$

$$0.12 = \frac{0.95 V_2^2}{2g} \Rightarrow V_2 = \frac{0.12 \times 2 \times 9.81}{0.95}$$

$$V_2^2 = \frac{2.3544}{0.95} = 2.4783$$

$$V_2 = 1.439 \text{ m/s}$$

$$Q = A_2 V_2 = 0.01767 \times 1.574 = 0.02782 \text{ m}^3/\text{s} \Rightarrow 27.82 \text{ lit/s}$$

$$V_2 = 1.574 \text{ m/s}$$

$$A_2 = \frac{\pi}{4} (0.15)^2 = 0.01767 \text{ m}^2 \Rightarrow Q = A_2 V_2 = 0.01767 \times 1.439 = 0.02545 \text{ m}^3/\text{s} \Rightarrow 25.45 \text{ lit/s}$$

# Pitot tube

Difference between.

- Piezometer :- to find static pressure  $\Rightarrow$  when a fluid in rest condition.
- Pitot tube :- to find stagnation pressure.

Stagnation Pressure :- The pressure a fluid has when it is brought to rest isentropically (without energy loss due to friction/heat).

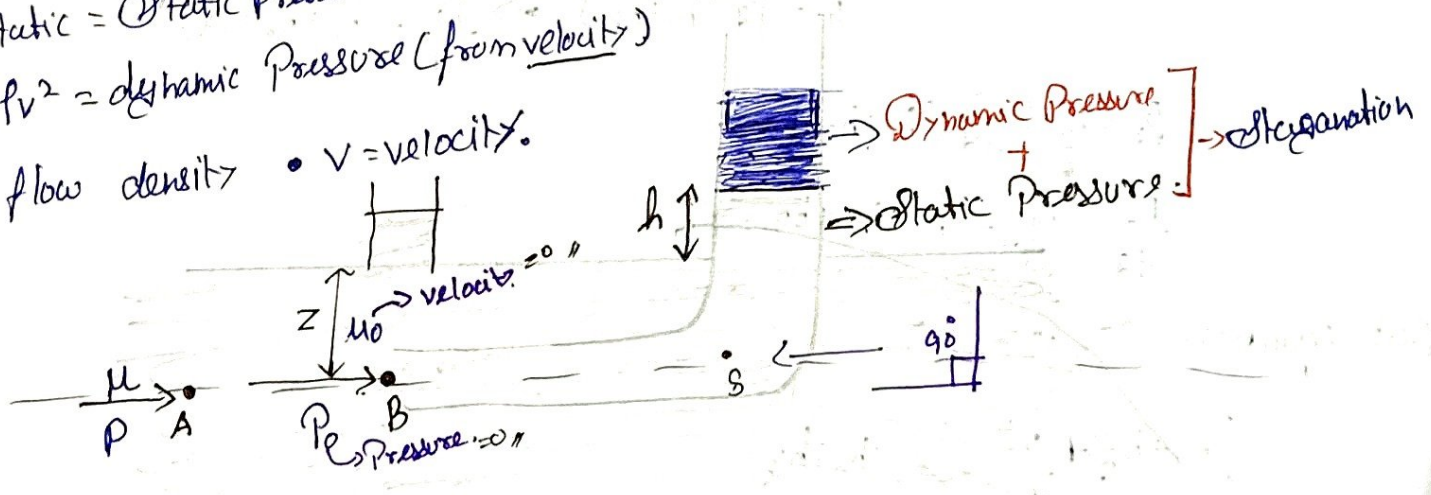
It is the sum of static pressure and dynamic pressure.

$$P_{\text{stagnation}} = P_{\text{static}} + \frac{1}{2} \rho v^2$$

$P_{\text{static}} = \text{Static Pressure}$

$\frac{1}{2} \rho v^2 = \text{dynamic Pressure (from velocity)}$

$\rho = \text{flow density}$  •  $v = \text{velocity}$



left  $\Rightarrow$  kinetic energy  $\Rightarrow$  pressure energy  $\Rightarrow$  right

ing point A to B on Bernoules eqn

$$\frac{v_1^2}{2g} + z_A = \frac{P_1}{\rho g} + \frac{v_2^2}{2g} + z_B$$

$\hookrightarrow$  same line

$$\frac{u^2}{2g} = \frac{P_0}{\rho g} + 0$$

$$\frac{u^2}{2g} = \frac{P_0}{\rho g} - \frac{P}{\rho g}$$

$$\frac{u^2}{2g} = (z+h) - z$$

$$P = \rho g z$$

$$P_0 = \rho g (z+h)$$

$$\frac{u^2}{2g} = h$$

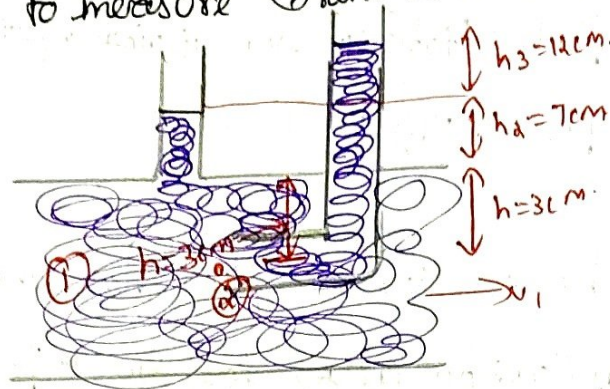
$$C_{th} = \sqrt{2gh} \Rightarrow \text{velocity of center of tube}$$

$$V_{\text{real}} = C_u = C \sqrt{2gh} \text{ (or)} \int_{C_u} \sqrt{2gh}$$

$\frac{\rho u^2}{2} = P_0 - P = P_{\text{stagnant}}$   
 $\downarrow$   
 Dynamic Press at a point

## Gate Problem:-

A Piezometer and a Pitot tube are tapped into a horizontal water pipe, as shown in figure, to measure static and stagnation (static + dynamic) pressure for the.



$$\textcircled{1} \rightarrow P_1$$

$$\textcircled{2} \rightarrow P_2$$

by apply Bernoullies eq<sup>n</sup> (1) & (2)

$$\frac{P_1}{\rho g} + \frac{v_1^2}{2g} + z_1 = \frac{P_2}{\rho g} + \frac{v_2^2}{2g} + z_2 \Rightarrow \textcircled{a}$$

by using hydrostatic eq<sup>n</sup>:-

$$P = \rho g h$$

$$P_1 = \rho g (h_1 + h_2) \Rightarrow \frac{P_1}{\rho g} = h_1 + h_2 \Rightarrow \textcircled{1}$$

$$P_2 = P_0 = \rho g (h_1 + h_2 + h_3)$$

$$\frac{P_2}{\rho g} - \frac{P_0}{\rho g} = h_1 + h_2 + h_3 \Rightarrow \textcircled{2}$$

Put equation (1) & (2) on (a)

$$\left( \frac{P_1}{\rho g} + h_2 \right) + \frac{v_1^2}{2g} = h_1 + h_2 + h_3$$

$$v_1 = \sqrt{2gh_3}$$

$$v_1 = \sqrt{2 \times 9.81 \times 0.12}$$

$$v_1 = \sqrt{2 \times 9.81 \text{ m/s}^2 \times 0.12 \text{ m}}$$

$$v_1 = \sqrt{2 \times 9.81 \times 0.12} \text{ (m/s)}$$

$$v_1 = 1.5342 \text{ m/s}$$

# Orifice meter

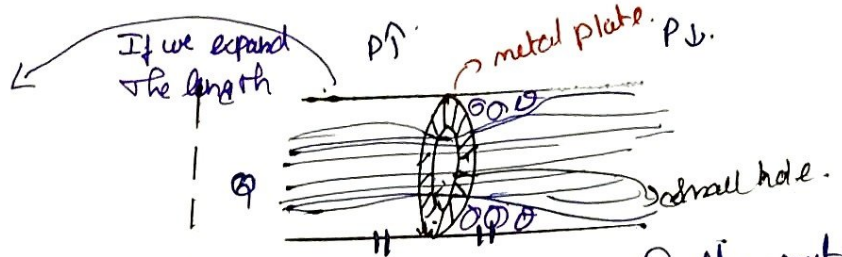
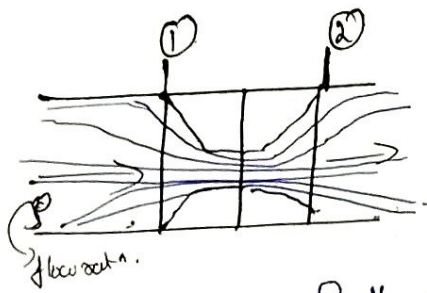
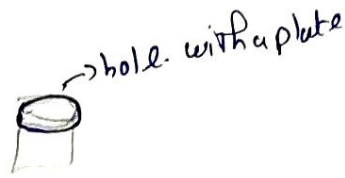
Q

⇒ to calculate flow rate through pipe.

⇒ orifice is like a tank opening.

⇒ a flow measuring instrument that uses a sharp-edged hole in a plate to measure the discharge (flow rate) in pipe.

⇒ to understanding deeply we want to compare venturi meter & orifice meter.



⇒ we know where the flow rate is minimum ⇒ we don't know where the flow rate is minimum.

⇒  $C_d = 1$

⇒  $C_d = 0.6$  to  $0.8$

⇒ less loss

⇒ more loss.

⇒ boundary section is long so we can get recovered. more loss in pressure.

⇒ Due to turbulence more loss.

⇒ Expensive, bulky, req more space.

⇒ cheap easy to install.

⇒ accurate, low energy loss, but costly & large.

⇒ simple, cheap but more energy loss

•  $Q_{th} = \frac{a_1 a_2 \sqrt{2gh}}{\sqrt{a_1^2 - a_2^2}}$

↑ Pipe section  
↑  $A_1, A_2$

•  $Q_{th} = \frac{a_1 a_0 \sqrt{2gh}}{\sqrt{a_1^2 - a_2^2}}$

↑ Pipe section  
↑ small hole

Coefficient of contraction =  $\boxed{C_c = \frac{A_2}{A_0}}$  ⇒ flow rate is minimum.

If  $a_0$  not given in question use this formula

# Momentum Balance

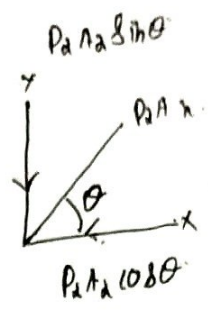
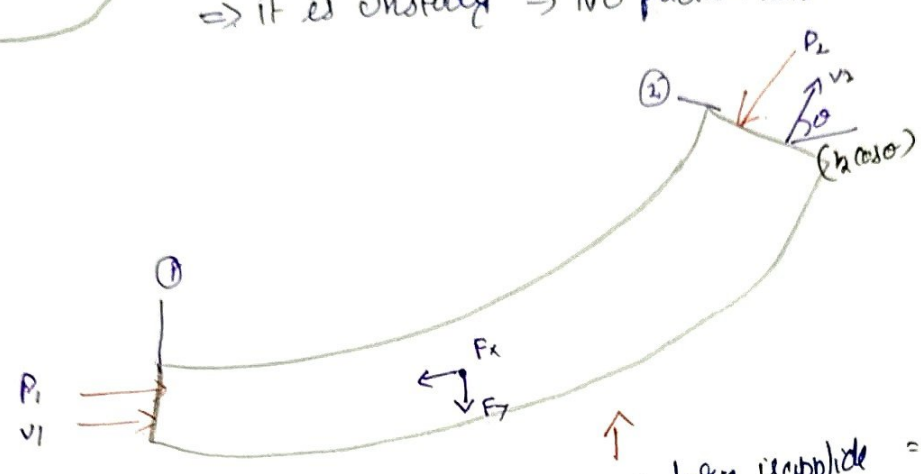
(SIOM)

(11)

ok  
A  
at  
ti  
oi  
is



⇒ what is total force acting on bending pipe.  
⇒ it is unsteady ⇒ no friction loss.



$$\sum F = m \cdot a$$

- $F_{net}$  = total (net) force acting on object.
- $m$  = mass of the object.
- $a$  ⇒ acceleration produced.

- If more force is applied = more acceleration.
- If mass is bigger = acceleration will be smaller.
- $a = F/m = 10/10 = 1 \text{ m/s}^2$

$$\sum F = \dot{m} \cdot (\Delta V)$$

*(mass flow rate)*

- $\Delta V$  = change in velocity ( $v_2 - v_1$ )
- $\dot{m}$  = mass flow rate.
- $\sum F$  = net force.

$\sum F = \dot{m} (v_2 - v_1)$  ⇒ constant ⇒ impulse momentum equation.

In x-direction:-  

$$P_1 A_1 - F_x - P_2 A_2 \cos \theta = m (v_2 \cos \theta - v_1)$$

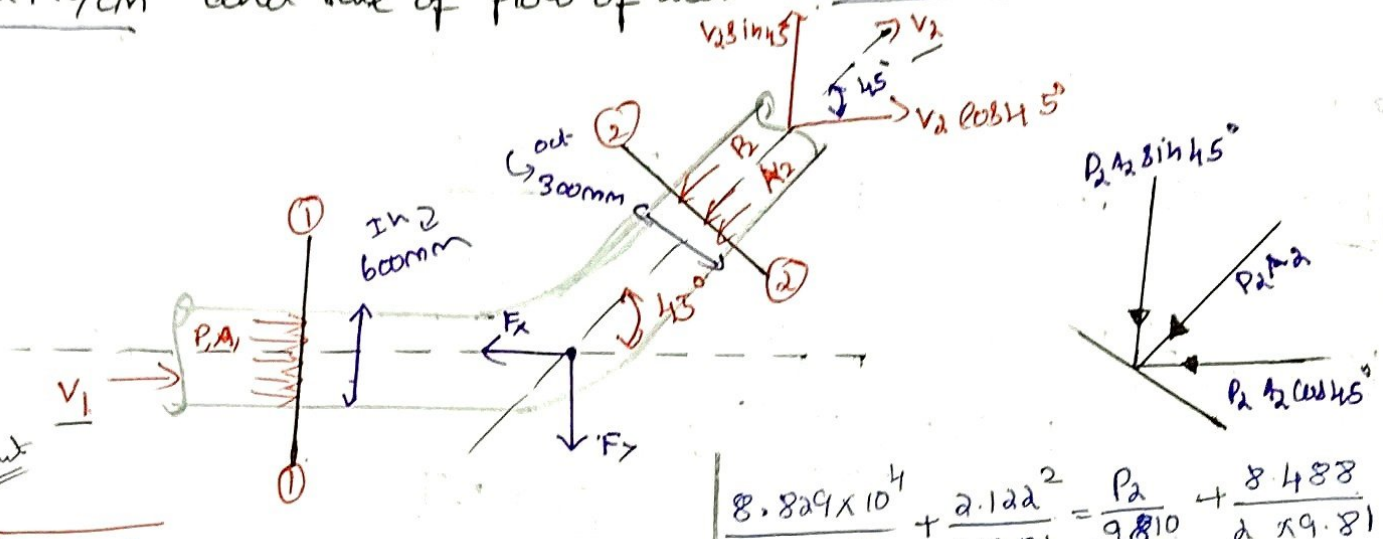
In y-direction:-  

$$- P_2 A_2 \sin \theta - F_y = m (v_2 \sin \theta - 0)$$
*(no y value)*

by :- 
$$F_R = \sqrt{F_x^2 + F_y^2}$$

Q.1

A 45° reducing bend is connected in a pipe line, the diameter at the inlet and the outlet of the bend 600mm and 300mm respectively. Find the magnitude and direction of the force exerted by water on the bend if the intensity of pressure at the inlet to bend is 8.829 N/cm<sup>2</sup> and rate of flow of water is 600 lit/s



Resultant  
 $F_R = \sqrt{F_x^2 + F_y^2}$

X-direction:-

$\sum F = \dot{m} (V_{out} - V_{in})$

$P_1 A_1 - P_2 A_2 \cos 45^\circ - F_x = \dot{m} (v_2 \cos 45^\circ - v_1) \rightarrow (1)$

Y-direction:-

$-P_2 v_2 \sin \theta - F_y = \dot{m} (v_2 \sin 45^\circ - 0) \rightarrow (2)$

$P_1 = 8.829 \text{ N/cm}^2 \Rightarrow 8.829 \times 10^4 \text{ N/m}^2$

$P_2 = ? ; Q = 0.6 \text{ m}^3/\text{sec} \approx 600 \text{ lit/s}$

$Q = A_1 v_1 = A_2 v_2$

$v_1 = \frac{Q}{A_1} \Rightarrow \frac{0.6}{0.2827} = 2.122 \text{ m/s}$   
 $A = \frac{\pi}{4} \times 0.6^2 = 0.2827$

$v_2 = \frac{Q}{A_2} = \frac{0.6}{0.07068} = 8.488 \text{ m/s}$   
 $A = \frac{\pi}{4} \times 0.3^2 = 0.07068$

by using Bernoulli's equation.

$\frac{P_1}{\rho g} + \frac{v_1^2}{2g} + z_1 = \frac{P_2}{\rho g} + \frac{v_2^2}{2g} + z_2$

$\frac{8.829 \times 10^4}{1000 \times 9.81} + \frac{2.122^2}{2 \times 9.81} = \frac{P_2}{9810} + \frac{8.488^2}{2 \times 9.81}$   
 $9 + 0.2295 = \frac{P_2}{9810} + 3.672$

$\frac{P_2}{9810} = 9.2295 - 3.672$   
 $= 5.5575 \text{ m of water}$

$P_2 = 5.5575 \times 9810 \text{ N/m}^2$

$P_2 = 5.45 \times 10^4 \text{ N/m}^2$

$m = \rho A v = \rho Q$

$F_x = \rho Q [v_1 - v_2 \cos \theta] + P_1 A_1 - P_2 A_2 \cos \theta$   
 $= 1000 \times 0.6 [2.122 - 8.488 \cos 45^\circ] + 8.829 \times 10^4 \times 0.2827 - 5.45 \times 10^4 \times 0.07068 \times \cos 45^\circ$   
 $\Rightarrow -2327.9 + 24959.6 - 2720.3 \Rightarrow 24959.6 - 5048.2$   
 $\Rightarrow 19911.4 \text{ N}$

$F_y = \rho Q [-v_2 \sin \theta] - P_2 A_2 \sin \theta$   
 $\Rightarrow 1000 \times 0.6 [-8.488 \sin 45^\circ] - 5.45 \times 10^4 \times 0.07068 \times \sin 45^\circ$   
 $\Rightarrow -3601.1 - 2721.1 \Rightarrow -6322.2 \text{ N}$

Resultant force:  $F_R = \sqrt{F_x^2 + F_y^2}$   
 $= \sqrt{(19911.4)^2 + (-6322.2)^2}$   
 $\Rightarrow 20890.9 \text{ N}$

The angle made by resultant force:  
 $\tan \theta = \frac{F_y}{F_x} = \frac{6322.2}{19911.2} = 0.3175$   
 $\theta = \tan^{-1}(0.3175)$   
 $\theta = 17^\circ 36'$

# NAVIER-STOKES EQUATION

by general term Newton's law applicable only (solid particle)

$$F = m \times a$$

• Navier-Stokes are the scientist who modify this equation for (liquids)

$$\text{kg (Newton-2)} = m = \frac{m}{v} = \rho \rightarrow \text{mass [m]}$$

FORCE [F] :-

$$F = F_g + F_p + F_v + F_T + F_c \Rightarrow \text{General equation of motion.}$$

↑ Gravity      ↑ velocity      ↑ Compressibility  
 ↓ Pressure      ↓ Turbulent

• Let us want to consider incompressible flow - [omit  $F_c$ ]

$$F = F_g + F_p + F_v + F_T \Rightarrow \text{Reynold's Equation}$$

• [omit  $F_T$ ]  $F = F_g + F_p + F_v \Rightarrow \text{Navier-Stokes Equation} \rightarrow \textcircled{2}$

⇒ consider Ideal fluid [omit  $F_v$ ] 'acceleration' - (Direction)  $\rightarrow \textcircled{3}$

$$\left( -\frac{\gamma v}{\gamma T} + v + \nabla v \right)$$

analogous [mass x acceleration]

$$F = F_g + F_p \Rightarrow \text{Euler's Equation}$$

by differential term:-

$$\frac{dP}{\rho} + g dz + v dv = 0$$

by Integration

$$\int \frac{dP}{\rho} + \int g dz + \int v dv = 0$$

$$\frac{P}{\rho} + gz + \frac{v^2}{2} = \text{constant}$$

$$\frac{P}{\rho g} + z + \frac{v^2}{2g} = \text{constant} \quad \text{Bernoulli's}$$

$$\frac{P_1}{\rho g} + z_1 + \frac{v_1^2}{2g} = \frac{P_2}{\rho g} + z_2 + \frac{v_2^2}{2g} + h_L \quad \text{Eq}$$

losses

The formula of Navier-Stokes equation

